

Research Article

A Coupling Electromechanical Inhomogeneous Cell-Based Smoothed Finite Element Method for Dynamic Analysis of Functionally Graded Piezoelectric Beams

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Received 16 April 2019; Accepted 7 July 2019; Published 6 August 2019

Guest Editor: Ayman E. Hamada

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To accurately simulate the continuous property change of functionally graded piezoelectric materials (FGPMs) and overcome the overstiffness of the finite element method (FEM), we present an electromechanical inhomogeneous cell-based smoothed FEM (ISFEM) of FGPMs. Firstly, ISFEM formulations were derived to calculate the transient response of FGPMs, and then, a modified Wilson- θ method was deduced to solve the integration of the FGPM system. The true parameters at the Gaussian integration point in FGPMs were adopted directly to replace the homogenization parameters in an element. ISFEM provides a close-to-exact stiffness of the continuous system, which could automatically and more easily generate for complicated domains and thus significantly decrease numerical errors. The accuracy and trustworthiness of ISFEM were verified as higher than the standard FEM by several numerical examples.

1. Introduction

Because of their outstanding electromechanical properties, easy fabricability, and preparation flexibility, piezoelectric materials are extensively applied as sensors and actuators to monitor and modulate the response of structures [1, 2]. Piezoelectric actuators and sensors are innovative for microscopic electromechanical systems and intelligent material systems, particularly in aerospace and medical fields [3]. Conventional piezoelectric sensors and actuators comprise multiple layers of various piezoelectric materials [4–7]. Moreover, piezoelectric layers with uniform material properties are limited by large bending displacement, stress concentration, creeping at high temperature, and failure from interfacial unbounding. All these phenomena are induced by mechanical or electric loading at layer interfaces [8].

To overcome the above limitations, Zhu et al. introduced and fabricated functionally graded piezoelectric material (FGPM) sensors and actuators [9, 10]. FGPMs change nonstop in one or more directions without generating internal stress concentration despite the production of large displacements. Takagi et al. fabricated FGPM bimorph actuators by using a mixed system of lead-zirconate-titanate (PZT) and Pt [11]. Nowadays, FGPMs are widely used intelligent materials for sensors and actuators in microstructural engineering. Many efforts have been made to analyze the behaviors and static/dynamic responses of FGPMs (e.g., shells, beams, and plates), such as the wave propagation study of FGPM plates based on the laminate theory [12]. Moreover, exact 3D analysis of FGPM rectangular plates was conducted by using a state-space approach [13] and to investigate the natural frequencies and mode shapes after being poled perpendicular to the middle plane [14]. The above method was also applied to explore the free vibration of rectangular FGPM plates [15]. The semianalytical finite element method (FEM) was used to investigate the static response of anisotropic and linear functionally graded magneto-electro-elastic plates [16]. Lezgy-Nazargah et al. [17] carried out static and dynamic

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analyses on piezoelectric beams by using a refined sinus model. The results have been found in good agreement with the reference solutions for various electrical and mechanical constrained conditions. Meanwhile, the 3D exact state-space solution [18] and Peano series solution [19] were developed for the cylindrical bending vibration of the FGPM laminates, respectively. Layerwise FEM was adopted to investigate the displacement and stress responses of an FGPM bimorph actuator [20]. Qiu et al. inhibited the vibration of a smart flexible clamped plate by using piezoelectric ceramic patch sensors and actuators [21]. The Timoshenko beam theory was used to analyze the static and dynamic responses of FGPM actuators to thermo-electro-mechanical loading [22]. The first-order shear deformation theory was used to study the static bending, free vibration, and dynamic responses of FGPM plates under electromechanical loading [23]. The free and induced vibrations of FGPM beams under thermoelectro-mechanical loading were characterized using the 3order shear deformation beam theory [24]. A high-order theory for FGPM shells was proposed based on the generalized Hamilton's principle [8]. Although the FEM (hversion) is adequate for low-frequency vibration analysis, it is not well suited to the vibration analysis of medium- or high-frequency regimes [25]. The spectral finite element method (SFEM) [26, 27] and the weak form quadrature method (QEM) [28, 29] are developed for the dynamic analysis of FGPM beams and structures.

Though FEM is the most widely used and effective numerical approach in practical issues in research and engineering (including mechanics of vibration), it is not necessarily fully perfect or cannot be further improved. For example, the probable overestimation of stiffness in solid structures may lead to locking behavior and inaccurate stress-solving [30]. By adding strain smoothing into FEM [31], Liu et al. established a series of cell-based [32-36], node-based [37, 38], edge-based [39-41], or face-based [42, 43] smoothed FEMs (S-FEMs) and their combinations [44–47]. These S-FEMs with different properties can be used to get desired solutions for a variety of benchmarks and practical mechanic issues [48-50]. The strain-smoothing operations can reduce or alleviate the overstiffness of standard FEM, significantly improving the accuracy of both primal and dual quantities [51]. Moreover, owing to absence of parametric mapping, the shape function derivatives and S-FEM models established in elasticity are not required to be insensitive to mesh distortion [52]. S-FEMs have been successfully extended to analyze the dynamic control of piezoelectric sensors and actuators, topological optimization of linear piezoelectric micromotors, statics, frequency, or defects of smart materials [53-63]. Zheng et al. [64] utilized the cell-based smoothed finite element method with the asymptotic homogenization method to analyze the dynamic issues on micromechanics of piezoelectric composite materials. Zhou et al. [65, 66] deduced the linear and nonlinear cell-based smoothed finite element method of functionally graded magneto-electro-elastic (MEE) structures and further examined the transient responses of MEE sensors or energy harvest structures considering the damping factors. However, there is little literature reported concerning the

dynamic response of FGPMs using the electromechanical inhomogeneous cell-based smoothed finite element method. Because of versatility, S-FEMs become convenient and efficient numerical approaches to address different physical issues.

Given the continuous change of the gradient of material properties along the thickness x_3 direction and with cellbased gradient smoothing, we deduced the basic formula of ISFEM and a modified Wilson- θ method to solve the integral solution of the FGPM system. The displacements and potentials of FGPM cantilever beams under sine wave load, cosine wave load, step wave load, and triangular wave load were analyzed in comparison with FEM.

2. Basic Equations for Piezoelectric Materials

2.1. Geometry and Coordinate System. Each beam has a rectangular uniform cross section and is made of N_i layers either completely or partially composed of FGPM beams. The Cartesian coordinate system (x_1, x_2, x_3) and geometric parameters are illustrated in Figure 1.

2.2. Constitutive Equations. At the k-th layer, 3D linear constitutive equations are polarized along its global co-ordinates as follows:

$$\begin{bmatrix} \sigma_{11} \\ \sigma_{22} \\ \sigma_{33} \\ \sigma_{23} \\ \sigma_{13} \\ \sigma_{12} \end{bmatrix}^{(k)} = \begin{bmatrix} c_{11} & c_{12} & c_{13} & 0 & 0 & 0 \\ c_{12} & c_{22} & c_{23} & 0 & 0 & 0 \\ c_{13} & c_{23} & c_{33} & 0 & 0 & 0 \\ 0 & 0 & 0 & c_{44} & 0 & 0 \\ 0 & 0 & 0 & 0 & c_{55} & 0 \\ 0 & 0 & 0 & 0 & 0 & c_{66} \end{bmatrix}^{(k)} \begin{bmatrix} \varepsilon_{11} \\ \varepsilon_{22} \\ \varepsilon_{33} \\ 2\varepsilon_{23} \\ 2\varepsilon_{13} \\ 2\varepsilon_{12} \end{bmatrix}^{(k)} \\ - \begin{bmatrix} 0 & 0 & \varepsilon_{31} \\ 0 & 0 & \varepsilon_{32} \\ 0 & 0 & \varepsilon_{33} \\ 0 & \varepsilon_{24} & 0 \\ \varepsilon_{15} & 0 & 0 \\ 0 & 0 & 0 \end{bmatrix}^{(k)} \begin{bmatrix} E_1 \\ E_2 \\ E_3 \end{bmatrix}^{(k)}, \\ \begin{bmatrix} D_1 \\ D_2 \\ D_3 \end{bmatrix}^{(k)} = \begin{bmatrix} 0 & 0 & 0 & 0 & \varepsilon_{15} & 0 \\ 0 & 0 & 0 & \varepsilon_{24} & 0 & 0 \\ \varepsilon_{31} & \varepsilon_{32} & \varepsilon_{33} & 0 & 0 & 0 \end{bmatrix}^{(k)} \begin{bmatrix} \varepsilon_{11} \\ \varepsilon_{22} \\ \varepsilon_{33} \\ 2\varepsilon_{23} \\ 2\varepsilon_{23} \\ 2\varepsilon_{13} \\ 2\varepsilon_{23} \\ 2\varepsilon_{13} \\ 2\varepsilon_{23} \\ 2\varepsilon_{13} \\ 2\varepsilon_{23} \\ 2\varepsilon_{13} \\ 2\varepsilon_{12} \end{bmatrix}^{(k)}$$
(2)
$$+ \begin{bmatrix} \chi_{11} & 0 & 0 \\ 0 & \chi_{22} & 0 \\ 0 & 0 & \chi_{33} \end{bmatrix}^{(k)} \begin{bmatrix} E_1 \\ E_2 \\ E_3 \end{bmatrix}^{(k)}, \\ \end{bmatrix}$$

where σ_{ij} and ε_{kl} are the stress tensor and infinitesimal strain tensor, respectively, E_i and D_i are electric field and electric



FIGURE 1: FGPM beams: Cartesian coordinate system and geometric parameters: length (L), width (b), and height (h).

displacement vector components, respectively; e_{ik} , c_{kl} , and X_{ij} are the piezoelectric, elastic, and dielectric material constants, respectively. Different from homogeneous piezoelectric materials, the three constants are dependent on coordinate x_3 . We assume that the material properties along the thickness direction are arbitrarily distributed as follows:

$$c_{ijkl}(x_3) = c_{ijkl}^0 f(x_3),$$

$$e_{ikl}(x_3) = e_{ikl}^0 f(x_3),$$

$$\chi_{ik}(x_3) = \chi_{ik}^0 f(x_3), \quad i, j, k, l == 1, 2, 3,$$

(3)

where $f(x_3)$ is an arbitrary function and c_{kl}^0 , e_{kl}^0 , and χ_{kl}^0 are values at the plane $x_3 = 0$.

In an FGPM beam, equations (1) and (2) are reduced to

$$\overline{\mathbf{\sigma}} = \mathbf{C}\overline{\mathbf{\epsilon}} - \overline{\mathbf{e}}\overline{\mathbf{E}},$$

$$\overline{\mathbf{D}} = \overline{\mathbf{e}}^{\mathrm{T}}\overline{\mathbf{\epsilon}} + \overline{\chi}\overline{\mathbf{E}},$$
(4)

where

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$$\overline{\boldsymbol{\sigma}} = \begin{bmatrix} \sigma_{11} \\ \sigma_{33} \\ \sigma_{13} \end{bmatrix}, \quad \overline{\mathbf{C}} = \begin{bmatrix} c_{11} & c_{13} & 0 \\ c_{31} & c_{33} & 0 \\ 0 & 0 & c_{55} \end{bmatrix}, \quad \overline{\mathbf{E}} = \begin{bmatrix} 0 & e_{31} \\ 0 & e_{33} \\ e_{15} & 0 \end{bmatrix}, \quad \overline{\mathbf{E}} = \begin{bmatrix} E_1 \\ E_3 \\ \vdots \\ \overline{\mathbf{D}} = \begin{bmatrix} D_1 \\ D_3 \\ \vdots \\ \vdots \\ \overline{\mathbf{\chi}} = \begin{bmatrix} \chi_{11} & 0 \\ 0 & \chi_{33} \end{bmatrix}. \quad (5)$$

2.3. Weak Formulation. The principle of virtual work for a piezoelectric medium of volume Ω and regular boundary surface Γ can be written as

$$\delta \Pi = \delta U - \delta W = -\int_{\Omega} \delta \overline{\mathbf{\epsilon}}^{\mathrm{T}} \overline{\mathbf{\sigma}} \, d\Omega + \int_{\Gamma} \delta \mathbf{u}^{\mathrm{T}} \mathbf{F}_{\mathrm{s}} \, d\Gamma + \int_{\Omega} \delta \mathbf{u}^{\mathrm{T}} \mathbf{F}_{\mathrm{v}} \, d\Omega - \int_{\Omega} \rho \delta \mathbf{u}^{\mathrm{T}} \ddot{\mathbf{u}} \, d\Omega + \int_{\Omega} \delta \overline{\mathbf{E}}^{\mathrm{T}} \overline{\mathbf{D}} \, d\Omega \quad (6) + \int_{\Gamma} \overline{Q} \delta \mathbf{\varphi} \, d\Gamma - \int_{\Omega} \overline{q} \delta \mathbf{\varphi} \, d\Omega = 0,$$

where \mathbf{F}_{s} , \mathbf{F}_{v} , \mathbf{u} , and $\boldsymbol{\varphi}$ are the vectors of surface force, mechanical body force, node displacements, and node electrical potentials, respectively; \overline{q} , \overline{Q} , and ρ are the electrical body charge, surface charge, and mass density, respectively; and δ is the virtual quantity.

3. Electromechanical ISFEM

The solving domain Ω is discretized into $n_{\rm p}$ elements, which contain $N_{\rm n}$ nodes; the approximation displacement $\overline{\mathbf{u}}$ and the approximation electrical potential $\overline{\boldsymbol{\varphi}}$ for the FGPM problem can be expressed as

$$\overline{\mathbf{u}} = \sum_{i=1}^{n_{\mathrm{p}}} N_{i}^{\mathrm{u}} u_{i} = \mathbf{N}_{\mathrm{u}} \mathbf{u},$$

$$\overline{\mathbf{\phi}} = \sum_{i=1}^{n_{\mathrm{p}}} N_{i}^{\varphi} \varphi_{i} = \mathbf{N}_{\varphi} \boldsymbol{\varphi},$$
(7)

where N_u and N_{φ} are the ISFEM displacement shape function and electrical potential shape function, respectively.

Four-node element is divided into four smoothing subdomains. Field nodes, edge smoothing nodes, center smoothing nodes and edge Gaussian points, the outer normal vector distribution, and the shape function values are shown in Figure 2.

At any point \mathbf{x}^k in the smoothing subdomains Ω_i^k , the smoothed strain $\overline{\mathbf{\epsilon}}(\mathbf{x}^k)$ and the smoothed electric field $\overline{E}(\mathbf{x}^k)$ are

$$\overline{\mathbf{\epsilon}}(\mathbf{x}^k) = \int_{\Omega_i^k} \mathbf{\epsilon}(\mathbf{x}) \Phi(\mathbf{x} - \mathbf{x}^k) d\Omega, \qquad (8)$$

$$\overline{\mathbf{E}}(\mathbf{x}^{k}) = \int_{\Omega_{i}^{k}} \mathbf{E}(\mathbf{x}) \Phi(\mathbf{x} - \mathbf{x}^{k}) d\Omega, \qquad (9)$$



FIGURE 2: Smoothing subcells and the values of shape functions (reproduced from Zheng et al. [64] under the Creative Commons Attribution License/public domain).

where $\boldsymbol{\varepsilon}(\mathbf{x})$ and $\mathbf{E}(\mathbf{x})$ are the strain and electric field in FEM, respectively, and $\Phi(\mathbf{x} - \mathbf{x}^k)$ is the constant function:

$$\Phi(\mathbf{x} - \mathbf{x}^{k}) = \begin{cases} \frac{1}{A_{i}^{k}}, & \mathbf{x} \in \Omega_{i}^{k}, \\ & & \\ 0, & \mathbf{x} \notin \Omega_{i}^{k}, \end{cases}$$
(10)

where

$$A_i^k = \int_{\Omega_i^k} d\Omega.$$
(11)

Substituting equation (10) into equations (8) and (9), then we have

$$\overline{\mathbf{\epsilon}}(\mathbf{x}^k) = \frac{1}{A_i^k} \int_{\Gamma_i^k} \mathbf{n}_u^k \mathbf{u} \, d\Gamma, \qquad (12)$$

$$\overline{\mathbf{E}}(\mathbf{x}^k) = \frac{1}{A_i^k} \int_{\Gamma_i^k} \mathbf{n}_{\varphi}^k \boldsymbol{\varphi} \, d\Gamma, \qquad (13)$$

where Γ_i^k is the boundary of Ω_i^k and \mathbf{n}_u^k and \mathbf{n}_{φ}^k are the outer normal vector matrixes of the smoothing domain boundary

$$\mathbf{n}_{u}^{k} = \begin{bmatrix} n_{x_{1}}^{k} & 0\\ 0 & n_{x_{3}}^{k}\\ n_{x_{3}}^{k} & n_{x_{1}}^{k} \end{bmatrix},$$

$$\mathbf{n}_{\varphi}^{k} = \begin{bmatrix} n_{x_{1}}^{k}\\ n_{x_{3}}^{k} \end{bmatrix}.$$
(14)

Equations (12) and (13) can be rewritten as

$$\overline{\mathbf{\epsilon}}(\mathbf{x}^{k}) = \sum_{i=1}^{n_{c}} \overline{\mathbf{B}}_{u}^{i}(\mathbf{x}^{k})\mathbf{u}_{i},$$

$$\overline{\mathbf{E}}(\mathbf{x}^{k}) = -\sum_{i=1}^{n_{c}} \overline{\mathbf{B}}_{\varphi}^{i}(\mathbf{x}^{k})\boldsymbol{\varphi}_{i},$$
(15)

where $n_{\rm e}$ is the number of smoothing elements.

$$\overline{\mathbf{B}}_{u}^{i}(\mathbf{x}^{k}) = \frac{1}{A_{i}^{k}} \int_{\Gamma^{k}} \begin{bmatrix} N_{i}^{*} n_{x_{1}}^{*} & 0 \\ 0 & N_{i}^{u} n_{x_{3}}^{k} \\ N_{i}^{u} n_{x_{3}}^{k} & N_{i}^{u} n_{x_{1}}^{k} \end{bmatrix} d\Gamma, \quad (16)$$

$$\overline{\mathbf{B}}_{\varphi}^{i}\left(\mathbf{x}^{k}\right) = \frac{1}{A_{i}^{k}} \int_{\Gamma^{k}} \begin{bmatrix} N_{i}^{\varphi} n_{x_{1}}^{k} \\ N_{i}^{\varphi} n_{x_{3}}^{k} \end{bmatrix} d\Gamma.$$
(17)

At the Gaussian point \mathbf{x}_b^{G} , equations (16) and (17) are

$$\overline{\mathbf{B}}_{u}^{i}(\mathbf{x}^{k}) = \frac{1}{A_{i}^{k}} \sum_{b=1}^{n_{b}} \begin{pmatrix} N_{i}^{u}(\mathbf{x}_{b}^{G})n_{x_{1}}^{k} & 0\\ 0 & N_{i}^{u}(\mathbf{x}_{b}^{G})n_{x_{3}}^{k}\\ N_{i}^{u}(\mathbf{x}_{b}^{G})n_{x_{3}}^{k} & N_{i}^{u}(\mathbf{x}_{b}^{G})n_{x_{1}}^{k} \end{pmatrix} l_{b}^{k},$$
(18)

$$\overline{\mathbf{B}}_{\varphi}^{i}\left(\mathbf{x}^{k}\right) = \frac{1}{A_{i}^{k}} \sum_{b=1}^{n_{b}} \begin{pmatrix} N_{i}^{\varphi}\left(\mathbf{x}_{b}^{\mathrm{G}}\right)n_{x_{1}}^{k} \\ \\ N_{i}^{\varphi}\left(\mathbf{x}_{b}^{\mathrm{G}}\right)n_{x_{3}}^{k} \end{pmatrix} l_{b}^{k},$$

where \mathbf{x}_{b}^{G} and l_{b}^{k} are the Gaussian point and the length of the smoothing boundary, respectively, and n_{b} is the total number of boundaries for each smoothing subdomain. As the shape function is linearly changed along each side of the smoothing subdomain, one Gauss point is sufficient for accurate boundary integration [30].

The essential distinction between ISFEM and FEM is that FEM needs to construct the shape function matrix of the element, while ISFEM only needs to use the shape function at the Gaussian point of the smoothing element boundary and does not require to involve the shape function derivatives. The above can reduce the continuity requirement of the shape function, and therefore, the accuracy and convergence of the method are improved.

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FIGURE 3: Geometry of the cantilever beam.

TABLE 1: Material constants.

Material	с ₁₁ (GPa)	с ₁₂ (GPa)	с ₁₃ (GPa)	<i>c</i> ₃₃ (GPa)	<i>c</i> ₄₄ (GPa)	ρ (kg/m ³)	e_{31} (C/m ²)	e_{33} (C/m ²)	e_{15} (C/m ²)	χ_{11}/χ_0	χ_{33}/χ_0	$\chi_0 \ (C/N \cdot m^2)$
PZT-4	139.0	77.80	74.30	115.0	25.60	7600	-5.20	15.10	12.70	1476.0	1301.0	$\begin{array}{c} 8.854 \times 10^{-12} \\ 8.854 \times 10^{-12} \end{array}$
PZT-5H	126.0	79.60	84.10	114.0	23.30	7500	-6.50	17.44	23.30	1468.9	1698.3	



FIGURE 4: Sine wave load.



FIGURE 5: Variations of the displacement u_3 and electrical potential φ at the loading point with time when the gradient parameter was n = -5. (a) Displacement. (b) Electrical potential.

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FIGURE 6: Variations of the displacement u_3 and electrical potential φ at the loading point with time when the gradient parameter was n = -1, 0, 1, and 5. (a) Displacement (n = -1). (b) Electrical potential (n = -1). (c) Displacement (n = 0). (d) Electrical potential (n = 0). (e) Displacement (n = 1). (f) Electrical potential (n = 1). (g) Displacement (n = 5). (h) Electrical potential (n = 5).

The dynamic model of the FGPM electromechanical system can be derived from the Hamilton principle in the following form:

$$\overline{\mathbf{M}}\ddot{\mathbf{q}} + \overline{\mathbf{K}}\mathbf{q} = \overline{\mathbf{F}},\tag{19}$$

where

$$\overline{\mathbf{M}} = \begin{bmatrix} \mathbf{M}_{uu} & 0 \\ 0 & 0 \end{bmatrix},
\overline{\mathbf{q}} = \begin{cases} \overline{\mathbf{u}} \\ \overline{\mathbf{\phi}} \end{cases},
\overline{\mathbf{K}} = \begin{bmatrix} \mathbf{K}_{uu} & \mathbf{K}_{u\varphi} \\ \mathbf{K}_{u\varphi}^{\mathrm{T}} & \mathbf{K}_{\varphi\varphi} \end{bmatrix},
\mathbf{q} = \begin{cases} \mathbf{u} \\ \mathbf{\phi} \end{cases},
\overline{\mathbf{F}} = \begin{cases} \mathbf{F} \\ \mathbf{Q} \end{cases},
\mathbf{M}_{uu} = \sum_{e} \mathbf{M}_{uu}^{e},$$
(20)

$$\mathbf{M}_{uu}^{e} = \text{diag}\{m_1, m_1, m_2, m_2, m_3, m_3, m_4, m_4\},\$$

where $m_i = p_i T$, A_i^k (*i*=1, 2, 3, 4) is the mass of the *i*-th smoothing element corresponding to node *i*, *T* is the smoothing element thickness, and p_i is the density of Gaussian integration point of the *i*-th smoothing subdomain:



FIGURE 7: Convergence rate of ISFEM.

$$\begin{split} \mathbf{K}_{\mathrm{uu}} &= \sum_{i=1}^{n_{\mathrm{c}}} \overline{\mathbf{B}}_{\mathrm{u}}^{i\mathrm{T}} \overline{\mathbf{C}} \overline{\mathbf{B}}_{\mathrm{u}}^{i} A_{i}^{k}, \\ \mathbf{K}_{\mathrm{u}\varphi} &= \sum_{i=1}^{n_{\mathrm{c}}} \overline{\mathbf{B}}_{\mathrm{u}}^{i\mathrm{T}} \overline{\mathbf{e}} \overline{\mathbf{B}}_{\varphi}^{i} A_{i}^{k}, \\ \mathbf{K}_{\varphi\varphi} &= -\sum_{i=1}^{n_{\mathrm{c}}} \overline{\mathbf{B}}_{\varphi}^{i\mathrm{T}} \overline{\chi} \overline{\mathbf{B}}_{\varphi}^{i} A_{i}^{k}, \\ \mathbf{F} &= \int_{\Omega} \mathbf{N}_{\mathrm{u}}^{\mathrm{T}} \mathbf{F}_{\mathrm{v}} d\Omega - \int_{\Gamma_{q}} \mathbf{N}_{\mathrm{u}}^{\mathrm{T}} \mathbf{F}_{\mathrm{s}} d\Gamma, \\ \mathbf{Q} &= \int_{\Omega} \mathbf{N}_{\varphi}^{\mathrm{T}} \overline{\mathbf{Q}} d\Omega + \int_{\Gamma_{q}} \mathbf{N}_{\varphi}^{\mathrm{T}} \overline{\mathbf{q}} d\Gamma, \end{split}$$
(21)

where $n_{\rm c} = n_{\rm p} \times n_{\rm e}$.



FIGURE 8: Comparison of computational efficiency.

TABLE 2: Displacement u_3 and electrical potential φ at the loading point with the gradient parameter n = -5, -1, 0, 1, and 5 of the FGPM beam based on PZT-4.

T:	Tiald an aighte	<i>n</i> =	-5	<i>n</i> =	-1	<i>n</i> =	= 0	<i>n</i> =	= 1	n	= 5
Time (s)	Field variables	ISFEM	FEM	ISFEM	FEM	ISFEM	FEM	ISFEM	FEM	ISFEM	FEM
0.0001	$u_3 \times 10^{-10} \mathrm{m}$	12.852	12.658	2.760	2.751	1.707	1.702	1.006	1.004	0.0756	0.0754
0.0001	$\varphi \times 10^{-2} \mathrm{V}$	11.848	11.912	0.442	0.427	0.212	0.207	0.0961	0.0944	0.00112	0.00110
0.0004	$u_3 \times 10^{-9} \mathrm{m}$	28.517	28.367	4.735	4.715	2.855	2.842	1.738	1.731	0.187	0.187
0.0004	$\varphi \times 10^{-1} \mathrm{V}$	5.851	5.747	0.597	0.582	0.349	0.339	0.193	0.188	0.0805	0.782
0.0025	$u_3 \times 10^{-8} \mathrm{m}$	13.345	13.191	1.538	1.528	0.917	0.913	0.564	0.561	0.0882	0.0873
0.0023	φ (V)	2.447	2.391	0.198	0.191	0.112	0.109	0.0621	0.0605	0.00386	0.00372
0.004	$U_3 \times 10^{-9} \mathrm{m}$	77.799	77.682	10.154	10.146	6.165	6.123	3.727	3.725	0.514	0.516
0.004	φ (V)	1.434	1.406	0.129	0.125	0.0756	0.0740	0.0414	0.0401	0.00224	0.00221

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FIGURE 9: Mesh of the FGPM beam.

TABLE 3: Displacement u_3 and electrical potential φ at the loading point with the gradient parameter n = -5, -1, 0, 1, and 5 of the FGPM beam based on PZT-5H.

Time (s)	Field warishlas	<i>n</i> =	-5	<i>n</i> =	-1	n=	= 0	n	= 1	n	n = 5	
Time (s)	Field variables	ISFEM	FEM	ISFEM	FEM	ISFEM	FEM	ISFEM	FEM	ISFEM	FEM	
0.0001	$u_3 \times 10^{-10} \mathrm{m}$	13.691	13.424	2.951	2.942	1.826	1.821	1.076	1.074	0.0802	0.0800	
0.0001	$\varphi \times 10^{-2} \mathrm{V}$	12.098	12.493	0.413	0.403	0.190	0.194	0.0825	0.0843	0.000624	0.000649	
0.0004	$u_3 \times 10^{-9} \mathrm{m}$	31.394	31.196	5.583	5.561	3.381	3.367	2.050	2.042	0.207	0.206	
0.0004	$\varphi imes 10^{-1} \mathrm{V}$	5.901	6.057	0.599	0.607	0.349	0.350	0.192	0.190	0.0736	0.0735	
0.0025	$u_3 \times 10^{-8} \mathrm{m}$	16.122	16.006	2.017	2.003	1.118	1.117	0.741	0.736	0.107	0.106	
0.0025	φ (V)	2.522	2.584	0.219	0.219	0.122	0.123	0.0691	0.0689	0.00392	0.00393	
0.004	$u_3 \times 10^{-9} \mathrm{m}$	76.019	75.052	10.339	10.264	6.264	6.261	3.795	3.768	0.501	0.496	
0.004	φ (V)	1.402	1.437	0.113	0.113	0.0649	0.0653	0.0351	0.0350	0.00179	0.00178	



FIGURE 10: Cosine wave load.

TABLE 4: Displacement u_3 and electrical potential φ at the loading point with the gradient parameter n = -5, -1, 0, 1, and 5 of the FGPM beam based on PZT-4 under cosine load.

Time o (o)	Field meniables	<i>n</i> =	-5	<i>n</i> =	-1	n	= 0	n=	= 1	n=	= 5
Time (s)	Field variables	ISFEM	FEM	ISFEM	FEM	ISFEM	FEM	ISFEM	FEM	ISFEM	FEM
0.0001	$u_3 \times 10^{-9} \mathrm{m}$	20.427	20.159	4.387	4.382	2.713	2.710	1.600	1.599	0.120	0.120
0.0001	$\varphi \times 10^{-2} \mathrm{V}$	188.326	189.821	7.029	6.896	3.369	3.269	1.528	1.5044	0.0178	0.0175
0.0004	$u_3 \times 10^{-8} \mathrm{m}$	21.208	21.234	2.738	2.743	1.610	1.611	1.006	1.004	0.141	0.142
0.0004	$\varphi \times 10^{-1} \mathrm{V}$	27.889	27.949	3.315	3.216	0.193	0.187	1.113	1.098	0.0648	0.0639
0.0025	$u_3 \times 10^{-9} \mathrm{m}$	-33.487	-32.845	-12.058	-11.952	-4.445	-4.345	-4.436	-4.392	-0.225	-0.221
0.0023	φ (V)	-0.138	-0.137	-0.132	-0.128	-0.0491	-0.0479	-0.0486	-0.0470	-0.00086	-0.00085
0.004	$u_3 \times 10^{-9} \mathrm{m}$	-11.111	-11.501	-0.984	-0.969	-5.549	-5.616	-0.351	-0.339	-0.0981	-0.0913
0.004	φ (V)	-1.408	-1.365	-0.0176	-0.0174	-0.0653	-0.0635	0.00234	0.00229	0.000596	0.000579

TABLE 5: Displacement u_3 and electrical potential φ at the loading point with the gradient parameter n = -5, -1, 0, 1, and 5 of the FGPM beam based on PZT-5H under cosine load.

Time (a)	Field meniables	n =	-5	<i>n</i> =	-1	n=	= 0	n=	= 1	n=	= 5
Time (s)	Fleid variables	ISFEM	FEM	ISFEM	FEM	ISFEM	FEM	ISFEM	FEM	ISFEM	FEM
0.0001	$u_3 \times 10^{-9} \mathrm{m}$	2.902	2.901	4.691	4.685	2.902	2.901	1.710	1.710	0.128	0.127
0.0001	$\varphi \times 10^{-2} \mathrm{V}$	192.287	198.967	6.561	6.742	3.017	3.092	1.312	1.342	0.00992	0.0101
0.0004	$u_3 \times 10^{-8} \mathrm{m}$	23.941	23.948	3.486	3.493	2.069	2.064	1.281	1.284	0.160	0.159
0.0004	$\varphi imes 10^{-1} \mathrm{V}$	28.178	29.094	3.5841	3.613	2.108	2.121	1.203	1.201	0.0621	0.0616
0.0025	$u_3 \times 10^{-9} \mathrm{m}$	117.958	122.357	-4.831	-4.723	-7.090	-6.942	-1.777	-1.715	0.795	0.8136
0.0025	φ (V)	0.50863	0.5017	-0.0407	-0.0393	-0.0688	-0.0652	-0.0151	-0.0148	0.00338	0.00342
0.004	$u_3 \times 10^{-9} \mathrm{m}$	-183.26	-179.45	-27.847	-27.063	-9.636	-9.387	-10.233	-9.998	-1.287	-1.263
0.004	φ (V)	-2.273	-2.176	-0.284	-0.277	-0.0931	-0.0885	-0.0951	-0.0949	-0.00466	-0.00451

The application of the inhomogeneous smoothing element is to calculate stiffness matrix of the element. The parameters of four smoothing subdomains A_i^k (*i* = 1, 2, 3, 4) are various in the elements, so the actual parameters at the Gaussian integration point are taken directly in order to reflect the changes of material property in each element.

4. Modified Wilson- θ Method

The modified Wilson- θ method is an important scheme and an implicit integral way to solve the dynamic system equations [63]. If $\theta > 1.37$, the solution is unconditionally stable. The detailed procedures are showed as follows:

4.1. Initial Calculation

- (1) Formulate generalized stiffness matrix $\overline{\mathbf{K}}$, mass matrix $\overline{\mathbf{M}}$, and damping matrix $\overline{\mathbf{C}}$
- (2) Calculate initial values of \mathbf{q} , $\dot{\mathbf{q}}$, $\ddot{\mathbf{q}}$
- (3) Select the time step Δt and the integral constant θ (θ = 1.4)



FIGURE 11: Step wave load.

$$a_{0} = \frac{6}{(\theta \Delta t)^{2}},$$

$$a_{1} = \frac{3}{(\theta \Delta t)},$$

$$a_{2} = 2a_{1},$$

$$a_{3} = \frac{\theta \Delta t}{2},$$

$$a_{4} = \frac{a_{0}}{\theta},$$

$$a_{5} = -\left(\frac{a_{2}}{\theta}\right),$$

$$a_{6} = 1 - \frac{3}{\theta},$$

$$a_{7} = \frac{\Delta t}{2},$$

$$a_{8} = \frac{\Delta t^{2}}{6}.$$
(22)

- (4) Formulate an effective generalized stiffness matrix $\widetilde{\mathbf{K}}$: $\widetilde{\mathbf{K}} = \overline{\mathbf{K}} + a_0 \overline{\mathbf{M}} + a_1 \overline{\mathbf{C}}$
- 4.2. For Each Time Step
 - (1) Calculate the payload at time $t + \theta \Delta t$:

$$\widetilde{\mathbf{F}}_{t+\theta\Delta t}^{*} = \overline{\mathbf{F}}_{t} + \theta \left(\overline{\mathbf{F}}_{t+\Delta t} - \overline{\mathbf{F}}_{t}\right) + \overline{\mathbf{M}} \left(a_{0}\mathbf{q}_{t} + a_{2}\dot{\mathbf{q}}_{t} + 2\ddot{\mathbf{q}}_{t}\right)$$
(23)
$$+ \overline{\mathbf{C}} \left(a_{1}\mathbf{q}_{t} + 2\dot{\mathbf{q}}_{t} + a_{3}\ddot{\mathbf{q}}_{t}\right).$$

(2) Calculate the generalized displacement at time $t + \theta \Delta t$:

$$\widetilde{\mathbf{K}}\mathbf{q}_{t+\theta\Delta t} = \widetilde{\mathbf{F}}_{t+\theta\Delta t}^*.$$
 (24)

(3) Calculate the generalized acceleration, generalized speed, and generalized displacement at time $t + \Delta t$:

$$\begin{aligned} \ddot{\mathbf{q}}_{t+\Delta t} &= a_4 \left(\mathbf{q}_{t+\theta\Delta t} - \mathbf{q}_t \right) + a_5 \dot{\mathbf{q}}_t + a_6 \ddot{\mathbf{q}}_t, \\ \dot{\mathbf{q}}_{t+\Delta t} &= \dot{\mathbf{q}}_t + a_7 \left(\ddot{\mathbf{q}}_{t+\Delta t} + \ddot{\mathbf{q}}_t \right), \end{aligned} \tag{25}$$
$$\mathbf{q}_{t+\Delta t} &= \mathbf{q}_t + \Delta t \dot{\mathbf{q}}_t + a_8 \left(\ddot{\mathbf{q}}_{t+\Delta t} + 2 \ddot{\mathbf{q}}_t \right). \end{aligned}$$

5. Numerical Examples

Four numerical examples were conducted under sine wave load, cosine wave load, step wave load and triangular wave load, respectively. FGPM cantilever beams of the same dimensions (length L = 40 mm, width h = 5 mm and thickness b = 1 mm) were subjected to forced vibration (Figure 3). The material constants are shown in Table 1. And initial conditions were $\mathbf{q} = 0$ and $\dot{\mathbf{q}} = 0$ at t = 0 moment. The FGPM beams were made of PZT-4 or PZT-5H on basis of exponentially graded piezoelectric materials with the following material properties:

$$c_{kl}(x_3) = c_{kl}^0 f(x_3),$$

$$e_{kl}(x_3) = e_{kl}^0 f(x_3),$$

$$\chi_{kl}(x_3) = \chi_{kl}^0 f(x_3),$$

(26)

where $f(x_3) = e^{nx_3/h}$ and *n* is the gradient parameter.

5.1. Sine Wave Load. The load *F* applied to the free end and the load waveform is demonstrated in Figure 4. A convergence investigation with respect to meshes was first carried out. Four smoothing subcells were used for electromechanical ISFEM with $\Delta t = 1 \times 10^{-3}$ s. The variations of displacement u_3 and electrical potential φ at the loading point of the PZT-4-based FGPM beam combined with respect to time are shown in Figure 5. The results at n = -5 with the element number of 480, 800, 1200, or 1680 were compared with the reference solution [67]. The variations of u_3 and φ at the loading point combined with respect to time at n = -1, 0, 1, and 5 in comparison with the reference solution are shown in Figure 6 [67]. Figure 7 illustrates the total energy norm Err versus the mesh density at t = 0.002 s



FIGURE 12: Variations of displacement u_3 and electrical potential φ at the loading point with time of the FGPM beam based on PZT-4 when the gradient parameter was n = -5, 0 and 5 under step load. (a) Displacement. (b) Electrical potential.



FIGURE 13: Variations of displacement u_3 and electrical potential φ at the loading point with time of the FGPM beam based on PZT-5H when the gradient parameter was n = -5, 0 and 5 under step load. (a) Displacement. (b) Electrical potential.

and t = 0.01 s. The simulation results are well consistent among different numbers of meshes, which demonstrate the high convergence of ISFEM.

Figure 8 shows a comparison of calculation time between ISFEM and FEM at Intel [®] Xeon [®] CPU E3-1220 v3 [@] 3.10 GHz, 16GB RAM. The bandwidth of the system matrices for the FEM and ISFEM is identical. However, in the ISFEM, only the values of shape functions (not the derivatives) at the quadrature points are needed and the requirement of traditional coordinate transform procedure is not necessary to perform the numerical integration.



Therefore, the ISFEM generally needs less computational cost than the FEM for handling the dynamic analysis problems.

The variations of u_3 and φ at the loading point combined with t = 0.0001 s, 0.0004 s, 0.0025 s, and 0.004 s in the PZT-4based FGPM cantilever beam are shown in Table 2. The 80×10 meshes of ISFEM at n = -5, -1, 0, 1, and 5 are shown in Figure 9, and FEM is considered 160×20 elements. Clearly, the results of ISFEM with 80×10 elements are the same as the calculated results of FEM using 160×20 elements, suggesting ISFEM has higher accuracy.



FIGURE 14: Triangular wave load.



FIGURE 15: Variations of displacement u_3 and electrical potential φ at the loading point with time of the FGPM beam based on PZT-4 when the gradient parameter was n = -5, 0, and 5 under triangular load. (a) Displacement. (b) Electrical potential.

The variations of u_3 and φ at the loading point combined with time of the PZT-5H-based FGPM cantilever beam are listed in Table 3. Clearly, when *n* changes from -5 to 5, the maximum u_3 and φ decrease, which is consistent with the PZT-4-based FGPM cantilever beam. Furthermore, the results of ISFEM with 80×10 elements are the same as the calculated results of FEM using 160×20 elements, suggesting ISFEM has higher accuracy.

5.2. Cosine Wave Load. The cosine load F applied to the free end and load waveform is shown in Figure 10. The variations of u_3 and φ at the loading point combined with time of PZT-4- and PZT-5H-based FGPM cantilever beams are listed in Tables 4 and 5, respectively. The calculated results of ISFEM with 80 × 10 elements are the same as those of FEM using



5.3. Step Wave Load. The step load *F* applied to the free end and load waveform is indicated in Figure 11. The variations of u_3 and φ at the loading point combined with time of PZT-4- and PZT-5H-based FGPM cantilever beams are illustrated in Figures 12 and 13, respectively. It is clearly shown that ISFEM possesses higher accuracy than FEM for the calculated results of ISFEM using 80 × 10 elements are the same as FEM using 160 × 20 elements.

5.4. *Triangular Wave Load*. The triangular load *F* applied to the free end and load waveform is shown in Figure 14. The



FIGURE 16: Variations of the displacement u_3 and electrical potential φ at the loading point with time of the FGPM beam based on PZT-5H when the gradient parameter was n = -5, 0, and 5 under triangular load. (a) Displacement. (b) Electrical potential.

variations of u_3 and φ at the loading point combined with time of PZT-4- and PZT-5H-based FGPM cantilever beams are shown in Figures 15 and 16, respectively. It shows that the solutions of CS-FEM with less elements are the same as the solutions of FEM using more elements.

6. Conclusions

An electromechanical ISFEM was proposed given the continuous changes of the gradient of material properties along the thickness x_3 direction and with cell-based gradient smoothing. The modified Wilson- θ method was deduced to solve the integral solution of the FGPM system. The displacements and potentials of cantilever beams combining with sine load, cosine load, step load, and triangular load were analyzed by ISFEM in comparison with FEM.

- (1) ISFEM is correct and effective in solving the dynamic response of FGPM structures
- (2) ISFEM can reduce the systematic stiffness of FEM and provides calculations closer to the true values
- (3) ISFEM is more efficient than FEM and takes less computation time at the same accuracy

This study indicates a possibility to select suitable grading controlled by the power law index according to the application.

Data Availability

The data used to support the findings of this study are included within the article.

Conflicts of Interest

The authors declare no conflicts of interest.

Authors' Contributions

Liming Zhou and Bin Cai performed the simulations. Bin Cai contributed to the writing of the manuscript.

Acknowledgments

Special thanks are due to Professor Guirong Liu for the S-FEM Source Code in http://www.ase.uc.edu/~liugr/software.html. This work was financially supported by the National Key R&D Program of China (grant no. 2018YFF01012401-06), Jilin Provincial Department of Science and Technology Fund Project (grant no. 20170101043JC), Jilin Provincial Department of Education (grant nos. JJKH20180084KJ and JJKH20170788KJ), Fundamental Research Funds for the Central Universities, Science and Technological Planning Project of Ministry of Housing and Urban–Rural Development of the People's Republic of China (2017-K9-047) and Graduate Innovation Fund of Jilin University (grant no. 101832018C184).

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